Advanced Robot Control Dynamic linearisation

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Dynamic linearisation is based on similar concept to static linearisation. A set of h(q) functions is selected that cause regularity condition to **fail**.

The state vector is extended to accommodate for a new "dynamic" variable.

Finally, the dynamic decoupling matrix is calculated. The obtained matrix needs to be non-singular, thus providing additional constraints on the system.



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h(q) linearisation functions (1/2)

Let us select following linearising functions

$$h_1 = x + e\cos(\theta + \delta),$$

$$h_2 = y + e\sin(\theta + \delta).$$
 (1)

(3)

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Now, let us consider a condition for which the regularity condition was not satisfied, thus $\delta = \frac{\pi}{2}$. Then

$$h_1 = x + e\cos(\theta + \frac{\pi}{2}) = x - e\sin(\theta), \qquad (2)$$

$$h_2 = y + e\sin(\theta + \frac{\pi}{2}) = y + e\cos(\theta).$$



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h(q) linearisation functions (2/2)

The guidance point that lies on the unicycle axle was singular from perspective of static linearisation. However, for dynamic linearisation this is required condition.



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Unicycle kinematics (1/1)

Let us consider kinematic of unicycle in following shape

$$\begin{cases} \dot{x} = v \cos \theta \\ \dot{y} = v \sin \theta \\ \dot{\theta} = \omega \end{cases}$$
(4)







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New input to the system (1/2)

Inserting kinematics (4) into derivative of linearising functions (3) yields

$$\dot{h}_{1} = \dot{x} - \boldsymbol{e}\cos\theta\dot{\theta} = \boldsymbol{v}\cos\theta - \boldsymbol{e}\cos\theta\omega = \cos\theta(\boldsymbol{v} - \boldsymbol{e}\omega) \quad (5)$$

$$\dot{h}_2 = \dot{y} - e\sin\theta \dot{\theta} = v\sin\theta - e\sin\theta\omega = \sin\theta(v - e\omega)$$
 (6)

Let us consider

$$\begin{cases} \chi_1 = \mathbf{v} - \mathbf{e}\omega \\ \dot{\chi}_1 = \mathbf{w}_1 \end{cases}$$
(7)





New input to the system (2/2)

Then derivatives of the linearising functions hold following form

$$\begin{cases} \dot{h}_1 = \chi_1 \cos \theta \\ \dot{h}_2 = \chi_1 \sin \theta \end{cases}$$
(8)



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Extending state vector

Initially, the state vector was composed of position and orientation

$$\boldsymbol{q} = \begin{bmatrix} \boldsymbol{x} & \boldsymbol{y} & \boldsymbol{\theta} \end{bmatrix}^{T}.$$
 (9)

Extended state vector holds additional variable χ_1

$$\boldsymbol{q}_{\boldsymbol{\theta}} = \left[\begin{array}{ccc} \boldsymbol{x} & \boldsymbol{y} & \boldsymbol{\theta} & \boldsymbol{\chi}_1 \end{array} \right]^T. \tag{10}$$

9/37



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Dynamic decoupling matrix (1/2)

$$\ddot{h}_1 = \dot{\chi}_1 \cos\theta + \chi_1 (-\sin\theta) \dot{\theta} = \dot{\chi}_1 \cos\theta - \omega \chi_1 \sin\theta$$
(11)

$$\ddot{h}_2 = \dot{\chi}_1 \sin \theta + \chi_1 \cos \theta \dot{\theta} = \dot{\chi}_1 \sin \theta + \omega \chi_1 \cos \theta$$
(12)

Let $w_2 = \omega$ and since $w_1 = \dot{\chi}_1$ then above can be rewritten in matrix form

$$\begin{pmatrix} \ddot{h}_1\\ \ddot{h}_2 \end{pmatrix} = \begin{bmatrix} \cos\theta & -\chi_1 \sin\theta\\ \sin\theta & \chi_1 \cos\theta \end{bmatrix} \begin{pmatrix} w_1\\ w_2 \end{pmatrix}.$$
 (13)

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Dynamic decoupling matrix (2/2)

Further, it can be written as

$$\ddot{h} = K_d(q) w, \tag{14}$$

where K_d is known as dynamic decoupling matrix. Similarly to the regularity condition for static linearisation, it is required for K_d not to be a singular matrix and invertible, thus

$$\det K_d \neq 0. \tag{15}$$



Constraints (1/2)

For given linearising functions (1) the dynamic decoupling matrix has following shape

$$\mathcal{K}_{d} = \begin{bmatrix} \cos\theta & -\chi_{1}\sin\theta \\ \sin\theta & \chi_{1}\cos\theta \end{bmatrix}.$$
 (16)

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Calculation of the determinant of above matrix yields

det
$$K_d = \chi_1 \cos^2 \theta + \chi_1 \sin^2 \theta = \chi_1 \left(\cos^2 \theta + \sin^2 \theta\right) = \chi_1.$$
 (17)



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Constraints (2/2)

If we want to control dynamically linearised system

$$\ddot{h} = \mathcal{K}_{d}(q)u \tag{18}$$

then

$$\chi_1 \neq 0. \tag{19}$$

$$v - e\omega \neq 0.$$
 (20)

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The system must be constantly on the move. The dynamic linearisation is a good choice for trajectory tracking problems.



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Diffeomorphism

If a diffeomorphism $f: q_e \rightarrow \Phi$ exists between extended state q_e and linear variables Φ ,

$$\boldsymbol{q} = \begin{bmatrix} \boldsymbol{x} & \boldsymbol{y} & \boldsymbol{\theta} & \chi_1 \end{bmatrix}^T, \quad (21)$$

$$\Phi = \begin{bmatrix} h_1 & h_2 & \dot{h}_1 & \dot{h}_2 \end{bmatrix}^T m$$
(22)

3

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then there is no problem with tracking entire posture including position and **orientation** which was not possible with static linearisation.

Dynamic linearisation allows for full linearisation of state space including orientation.

Control low (1/2)

Given system

$$\ddot{h} = K_d(q)u \tag{23}$$

a following control low can be proposed

$$u = K_d^{-1}(\ddot{h}_d - K_1 \dot{e}_h - K_0 e_h), \qquad (24)$$

where K_0 and K_1 are positively defined matrices, and errors are $e_h = h - h_d$ and $\dot{e}_h = \dot{h} - \dot{h}_d$.



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Control low (2/2)

Let us close the control loop by injecting control low (24) into our system (23).

$$\ddot{h} = K_d(q) K_d^{-1} (\ddot{h}_d - K_1 \dot{e}_h - K_0 e_h).$$
(25)

$$\ddot{h} = \ddot{h}_d - K_1 \dot{\boldsymbol{e}}_h - K_0 \boldsymbol{e}_h. \tag{26}$$

$$\ddot{\boldsymbol{e}}_h + \boldsymbol{K}_1 \dot{\boldsymbol{e}}_h + \boldsymbol{K}_0 \boldsymbol{e}_h = \boldsymbol{0}. \tag{27}$$

Using Laplace transform

$$\mathscr{L}\{\ddot{\boldsymbol{e}}_h + \boldsymbol{K}_1 \dot{\boldsymbol{e}}_h + \boldsymbol{K}_0 \boldsymbol{e}_h\} = \boldsymbol{E}_h(\boldsymbol{s}) \left(\boldsymbol{s}^2 + \boldsymbol{K}_1 \boldsymbol{s} + \boldsymbol{K}_0\right). \tag{28}$$

Based on Hurwitz criteria, if $K_0, K_1 > 0$ then (28) is Hurwitz of Science and Technology



Real unicyle control signals (1/4)

Given following control law

$$\begin{pmatrix} w_1 \\ w_2 \end{pmatrix} = u = K_d^{-1} (\ddot{h}_d - K_1 \dot{e}_h - K_0 e_h),$$
(29)

we have

$$u = \begin{pmatrix} w_1 \\ w_2 \end{pmatrix} = \begin{pmatrix} \dot{\chi}_1 \\ \omega \end{pmatrix} = \begin{pmatrix} \dot{v} - e\dot{\omega} \\ \omega \end{pmatrix}.$$
 (30)



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Real unicyle control signals (2/4)

If $\dot{\chi}_1 = \dot{\textit{v}} - \textit{e}\dot{\omega}$ then

$$\dot{\chi}_1 = \dot{\boldsymbol{v}} - \boldsymbol{e}\dot{\omega} \tag{31}$$

. Above is a concrete function calculated based on (29).

$$\dot{\boldsymbol{v}} = \dot{\chi}_1 + \boldsymbol{e}\dot{\omega} \tag{32}$$

and then

$$\dot{v} = w_1 + e \frac{d}{dt} w_2. \tag{33}$$

To calculate *v* integration is required.

$$\mathbf{v} = \int \left(\mathbf{w}_1 + \mathbf{e} \frac{d}{dt} \mathbf{w}_2 \right), \tag{34}$$

18/37



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 $v=\int w_1+ew_2.$

Real unicyle control signals (3/4)

Integration of w_1 is required to calculate real control signals. It also introduces a **delay** into the system.



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Real unicyle control signals (4/4)

$\omega=\textit{w}_{2}$ is also a concrete function calculated based on control law (29).



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20/37

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Kinematic vehicle (1/2)



Kinematic vehicle (2/2)

Let us consider kinematic for the object

$$\begin{cases} \dot{x} = v \cos \theta \\ \dot{y} = v \sin \theta \\ \dot{\theta} = \frac{v}{7} \tan \phi \\ \dot{\phi} = \omega \end{cases}$$
(36)

where v is linear velocity, I is wheel base, while ϕ is attack angle of the steering wheel and ω is angular velocity of the rotating steering wheel.



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h(q) linearisation functions (1/6)

Let us consider straightforward linearisation functions

$$h_1 = x$$

$$h_2 = y. \tag{37}$$

The first derivatives are directly taken from kinematics.

$$\dot{h}_1 = v \cos \theta$$
 (38)

$$\dot{h}_2 = v \sin \theta$$
 (39)

Further differentiation yields



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h(q) linearisation functions (2/6)

$$\begin{aligned} \ddot{h}_1 &= \dot{v}\cos\theta - v\sin\theta\dot{\theta} \\ \ddot{h}_1 &= \dot{v}\cos\theta - v\sin\theta v\tan\phi \\ \ddot{h}_1 &= \dot{v}\cos\theta - \sin\theta\tan\phi v^2 \end{aligned} \tag{40}$$

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and for \ddot{h}_2

$$\begin{split} \ddot{h}_2 &= \dot{v}\sin\theta + v\cos\theta\dot{\theta}\\ \ddot{h}_2 &= \dot{v}\sin\theta + v\cos\theta v\tan\phi\\ \ddot{h}_2 &= \dot{v}\sin\theta + \cos\theta\tan\phi v^2. \end{split}$$



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h(q) linearisation functions (3/6)

Assuming that $\dot{v} = z$ is a new input to the system we have

$$\ddot{h}_1 = z\cos\theta - \sin\theta\tan\phi v^2, \qquad (42)$$

$$\ddot{h}_2 = z \sin \theta + \cos \theta \tan \phi v^2. \tag{43}$$

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Further differentiation reveals relationship between new control inputs and state.

$$\ddot{h}_{1} = \dot{z}\cos\theta - z\sin\theta\dot{\theta}$$
$$-\cos\theta\dot{\theta}\tan\phi v^{2} - \sin\theta\frac{1}{\cos^{2}\phi}\dot{\phi}v^{2} - \sin\theta\tan\phi 2v\dot{v} \qquad (44)$$

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h(q) linearisation functions (4/6)

Let
$$\dot{v} = z$$
 and $\dot{z} = w_1$, while $\dot{\phi} = \omega$ then

$$\ddot{h}_1 = w_1 \cos \theta - z \sin \theta \dot{\theta} - \cos \theta \dot{\theta} \tan \phi v^2 - \sin \theta \frac{1}{\cos^2 \phi} \omega v^2 - \sin \theta \tan \phi 2vz \qquad (45)$$

Also $\dot{\theta} = v \tan \phi$, therefore above can be further rewritten and grouped accordingly to control input

$$\ddot{h}_{1} = \cos\theta w_{1} - \sin\theta \frac{1}{\cos^{2}\phi} v^{2}\omega$$
$$-\cos\theta \tan^{2}\phi v^{3} - 3zv\sin\theta v \tan\phi \qquad (46)$$



h(q) linearisation functions (5/6)

$$\ddot{h}_{2} = \dot{z}\sin\theta + z\cos\theta\dot{\theta}$$

$$-\sin\theta\dot{\theta}\tan\phi v^{2} + \cos\theta\frac{1}{\cos^{2}\phi}\dot{\phi}v^{2} + \cos\theta\tan\phi 2v\dot{v} \quad (47)$$
Let $\dot{v} = z$ and $\dot{z} = w_{1}$, while $\dot{\phi} = \omega$ then
$$\vdots$$

$$\ddot{h}_{2} = w_{1} \sin \theta + z \cos \theta \dot{\theta}$$
$$-\sin \theta \dot{\theta} \tan \phi v^{2} + \cos \theta \frac{1}{\cos^{2} \phi} \omega v^{2} + \cos \theta \tan \phi 2 v z \quad (48)$$

Let us group above and use $\dot{\theta} = v \tan \phi$ relation



h(q) linearisation functions (6/6)

Finally, the dynamically linearised system can be written in matrix form.

$$\left(\begin{array}{c} \ddot{h}_{1} \\ \dot{h}_{2} \end{array}\right) = \mathcal{K}_{d} \left(\begin{array}{c} \mathbf{w}_{1} \\ \boldsymbol{\omega} \end{array}\right) + f, \tag{50}$$

where

$$\mathcal{K}_{d} = \begin{bmatrix} \cos\theta & -\sin\theta \frac{1}{\cos^{2}\phi} \mathbf{v}^{2} \\ \sin\theta & \cos\theta \frac{1}{\cos^{2}\phi} \mathbf{v}^{2} \end{bmatrix},$$
(51)

28/37

$$= \begin{pmatrix} -\cos\theta\tan^2\phi v^3 - 3vz\sin\theta\tan\phi \\ -\sin\theta\tan^2\phi v^3 + 3vz\cos\theta\tan\phi \end{pmatrix}$$

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Constraints (1/3)

The dynamic decoupling matrix has following shape

$$\mathcal{K}_{d} = \begin{bmatrix} \cos\theta & -\sin\theta \frac{1}{\cos^{2}\phi} \mathbf{v}^{2} \\ \sin\theta & \cos\theta \frac{1}{\cos^{2}\phi} \mathbf{v}^{2} \end{bmatrix},$$
(53)

(54)

29/37

Calculation of the determinant of above matrix yields

$$\det K_d = \cos^2 \theta \frac{1}{\cos^2 \phi} v^2 + \sin^2 \theta \frac{v^2}{\cos^2 \phi}$$
$$= \frac{v^2}{\cos^2 \phi} \left(\cos^2 \theta + \sin^2 \theta\right) = \frac{v^2}{\cos^2 \phi}.$$



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Constraints (2/3)

If we want to control dynamically linearised system

$$\ddot{h} = K_d(q)u + f$$
 (55)

then

$$\frac{v^2}{\cos^2\phi} \neq 0. \tag{56}$$

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Equivalently

$$\begin{cases} \mathbf{v} \neq \mathbf{0} \\ \cos^2 \phi \neq \mathbf{0} \to \phi \neq \frac{\pi}{2} + \mathbf{k}\pi \end{cases} .$$
 (57)



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Constraints (3/3)

$$v \neq 0.$$
 (58)

The system must be constantly on the move.

The dynamic linearisation is a good choice for trajectory tracking problems.

$$\phi \neq \frac{\pi}{2} + k\pi. \tag{59}$$

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31/37

The steering wheel cannot be perpendicular to the main axle.



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Control low (1/2)

Now, we have our system in following form

$$\ddot{h} = K_d u + f. \tag{60}$$

32/37

Let us propose a control law for (60).

$$u = K_d^{-1}(-f + \ddot{h}_d - K_2 \ddot{e}_h - K_1 \dot{e}_h - K_0 e_h), \qquad (61)$$

where K_0 , K_1 and K_2 are positively defined matrices, and errors are $e_h = h - h_d$, $\dot{e}_h = \dot{h} - \dot{h}_d$ and $\ddot{e}_h = \ddot{h} - \ddot{h}_d$. To determine if system is stable we need to apply control law (61) to our system (60).

$$\ddot{h} = K_d(q)K_d^{-1}(-f + \ddot{h}_d - K_2\ddot{e}_h - K_1\dot{e}_h - K_0e_h) + f.$$
(62)



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$$\ddot{e}_h + K_2 \ddot{e}_h + K_1 \dot{e}_h + K_0 e_h = 0.$$



Control low (2/2)

Using Laplace transform

$$\mathscr{L}\{\ddot{e}_h + K_2 \ddot{e}_h + K_1 \dot{e}_h + K_0 e_h\} = E_h(s) \left(s^3 + K_2 s^2 + K_1 s + K_0\right).$$
(65)

Based on Hurwitz criteria, if $K_0, K_1, K_2 > 0$ then (65) is Hurwitz polynomial, thus the system is exponentially stable.



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Real kinematic vehicle control signals (1/3)

Given following control law

$$\begin{pmatrix} w_1\\ \omega \end{pmatrix} = u = K_d^{-1} (\ddot{h}_d - K_2 \ddot{e}_h - K_1 \dot{e}_h - K_0 e_h), \qquad (66)$$

we have

$$\boldsymbol{u} = \begin{pmatrix} \boldsymbol{w}_1 \\ \boldsymbol{\omega} \end{pmatrix} = \begin{pmatrix} \dot{\boldsymbol{z}} \\ \boldsymbol{\omega} \end{pmatrix} = \begin{pmatrix} \ddot{\boldsymbol{v}} \\ \boldsymbol{\omega} \end{pmatrix}.$$
(67)



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34/37

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Real kinematic vehicle control signals (2/3)

$$\ddot{v} = w_1. \tag{68}$$

Above is a concrete function calculated based on (66). To calculate v integration is required.

$$v = \int \int w_1, \qquad (69)$$

Integration of w_1 is required to calculate real control signals. It also introduces a **delay** into the system.



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Real kinematic vehicle control signals (3/3)

$\omega=\textit{w}_{2}$ is also a concrete function calculated based on control law (66).



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36/37

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Literature (1/1)

C. C. de Wit, B. Siciliano, and G. Bastin. Theory of Robot Control. Springer-Verlag London, 1996.



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